

1 **m-NLP inference models using simulation and**
2 **regression techniques**

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9 **Key Points:**

- 10 • 3-D kinetic PIC simulations are used to simulate currents collected by m-NLP in
11 order to create a synthetic solution library
- 12 • Models to infer physical parameters from m-NLP measurements are constructed
13 and assessed on the basis of synthetic and in situ data sets
- 14 • Promising new approaches are identified to analyze m-NLP measurements based
15 on simulation and in-situ data

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Abstract

Current inference techniques for processing multi-needle Langmuir Probe (m-NLP) data are often based on adaptations of the Orbital Motion-Limited (OML) theory which relies on several simplifying assumptions. Some of these assumptions, however, are typically not well satisfied in actual experimental conditions, thus leading to uncontrolled uncertainties in inferred plasma parameters. In order to remedy this difficulty, three-dimensional kinetic particle in cell simulations are used to construct a synthetic data set, which is used to compare and assess different m-NLP inference techniques. Using a synthetic data set, regression-based models capable of inferring electron density and satellite potentials from 4-tuples of currents collected with fixed-bias needle probes similar to those on the NorSat-1 satellite, are trained and validated. The regression techniques presented show promising results for plasma density inferences with RMS relative errors less than 20 %, and satellite potential inferences with RMS errors less than 0.2 V for potentials ranging from -6 V to -1 V. The new inference approaches presented are applied to NorSat-1 data, and compared with existing state-of-the-art inference techniques.

1 Introduction

Langmuir probes are widely used to characterize space plasma and laboratory plasma. A variety of Langmuir probe geometries are being used, such as spherical (Bhattarai & Mishra, 2017), cylindrical (Hoang, Clausen, et al., 2018), and planar probes (Lira et al., 2019; Johnson & Holmes, 1990; Sheridan, 2010). Probes can be operated in sweep mode (Lebreton et al., 2006), harmonic mode (Rudakov et al., 2001), or fixed biased mode (Jacobsen et al., 2010), for different types of missions and measurements. Despite operational differences, all Langmuir probes consist of conductors exposed to plasma to collect current as a function of bias voltage. A common approach to infer plasma parameters from Langmuir probes is to sweep the bias voltage and produce a current-voltage characteristic, which can be analyzed using theories such as the Orbital Motion-Limited (OML) (Mott-Smith & Langmuir, 1926) theory, the Allen-Boyd-Reynolds (ABR) theory (Allen et al., 1957; Chen, 1965, 2003), and the Bernstein-Rabinowitz-Laframboise (BRL) theory (Bernstein & Rabinowitz, 1959; Laframboise, 1966) to obtain plasma parameters such as density, temperature, and satellite floating potential. The temporal and, on a satellite, the spatial resolution of Langmuir probe measurements are determined by the sweep time, which varies based on the mission's scientific need and available resources. Considering the orbital speed to be around 7500 m/s for a satellite in low Earth orbit (LEO), the spatial resolution of sweep bias Langmuir probe can vary from tens of meters, to kilometers, depending on the sweep frequency. In order to study the formation of density irregularities that scale from meters to tens of kilometers at high and low latitudes, a sampling frequency of near 1 kHz is required (Hoang, Røed, et al., 2018; Jacobsen et al., 2010). A solution, proposed by Jacobsen is to use multiple fixed biased needle probes (m-NLPs) to sample plasma simultaneously at different bias potentials in the electron saturation region (Jacobsen et al., 2010). This approach would eliminate the need for sweeping the bias voltage, and greatly increase the sampling rate of the instrument.

The first inference models for m-NLPs relied on the OML approximation, from which the current I_e collected by a needle probe in the electron saturation region is written as:

$$I_e = -n_e e A \frac{2}{\sqrt{\pi}} \sqrt{\frac{kT_e}{2\pi m_e}} \left(1 + \frac{e(V_f + V_b)}{kT_e} \right)^\beta, \quad (1)$$

where n_e is the electron density, A is the probe surface area, e is the elementary charge, k is Boltzmann's constant, T_e is the electron temperature, V_f is the satellite floating potential, V_b is the bias potential of the probe with respect to the satellite, and β is a parameter related to probe geometry, density, and temperature (Marholm & Marchand, 2020; Hoang, Røed, et al., 2018). Several assumptions were made in the derivation of this inference equation; such as the probe length must be much larger than the Debye length,

and the plasma is non-drifting. If these assumptions are valid, then $\beta = 0.5$, and as first suggested by Jacobsen, a set of m-NLPs can be used to infer the electron density independently of the temperature (Jacobsen et al., 2010). For a satellite in near-Earth orbit at altitudes ranging from 550 km to 650 km, we can expect a Debye length of around 2-50 mm, and an orbital speed of around 7500 m/s. A common length for m-NLP instrument used on small satellites is ~ 25 mm (Bekkeng et al., 2010; Hoang, Clausen, et al., 2018; Hoang et al., 2019), which is often comparable to, and sometimes smaller than the Debye length. In lower Earth orbit, ion thermal speeds are usually less than the orbital speed, while electron thermal speeds are usually higher than the orbital speed. Thus, the orbital speed is expected to mainly affect ion saturation region currents for Langmuir probes. However, electrons can only penetrate the ion rarefied wake region behind the probe as much as ambipolar diffusion permits (Barjatya et al., 2009). As a result, electron saturation currents are also influenced by an orbital speed. One consequence is that the $\beta = 0.5$ assumption does not hold in Eq. 1, and a better approximation for the current is obtained with β values between 0.5 and 1. For example, in a hot filament-generated plasma experiment, Sudit and Woods showed that β can reach 0.75 for a ratio between the probe length and the Debye length in the range of 1 to 3. For larger Debye lengths, they also observed an expansion of the probe sheath from a cylindrical shape into a spherical shape (Sudit & Woods, 1994). Ergun and co-workers showed that with a ram speed of 4300 m/s in their simulations, the current collected by a 40.8 cm needle probe is better approximated with Eq. 1 using a β value of 0.67 instead of 0.55 calculated in a stationary plasma (Ergun et al., 2021). In the ICI-2 sounding rocket experiment, β calculated from three 25 mm m-NLPs varied between 0.3 to 0.7 at altitudes ranging from 150 to 300 km (Hoang, Røed, et al., 2018). Simulation results by Marholm et al. showed that even a 50 mm probe at rest can be characterized by a $\beta \sim 0.8$ (Marholm et al., 2019), in disagreement with the OML theory. In practice, needle probes are mounted on electrically isolated and equipotential guards in order to attenuate end effects on the side to which they are attached. The distribution of the current collected per unit length is nonetheless not uniform along the probe, as more current is collected near the end opposite to the guard. A study by Marholm & Marchand showed that for a cylindrical probe length that is 10 times the Debye length, β is approximately 0.72. For a probe length that is 30 times the Debye length, β is approximately 0.62, and with a guard, this number is reduced to 0.58 (Marholm & Marchand, 2020). Although this number approaches 0.5, 30 times the Debye length is a stringent requirement for OML to be valid, and it is hardly ever fulfilled in practice. Experimentally, Hoskinson and Hershkowitz showed that even with a probe length 50 times the Debye length, β is approximately 0.6, and the density inference based on an ideal $\beta = 0.5$ is 25 % too high (Hoskinson & Hershkowitz, 2006). Barjatya estimated that even a 10% error in β (to 0.55) can result in a 30 % or more relative error in the calculated density based on the $\beta = 0.5$ assumption (Barjatya & Merritt, 2018). In what follows, we find that densities estimated using Eq. 1 assuming $\beta = 0.5$ are about three times larger than the known values used as input in our simulations, as illustrated in Sec. 2.4.1. This is consistent with findings in (Barjatya & Merritt, 2018; Guthrie et al., 2021), considering β calculated in our simulation is in the range of 0.75 to 1. Another approach proposed to account for the fact that β is generally different from 0.5, consists of determining the n_e , V_b , T_e and β , as adjustable parameters in nonlinear fits of measured currents as a function of voltages. This led to remarkable agreement with density measured using a radio frequency impedance probe on the international space station (Barjatya et al., 2009, 2013; Debchoudhury et al., 2021). This method was originally applied to a probe operated in sweep voltage mode, but it can be straightforwardly adapted to fixed bias m-NLP measurements (Barjatya et al., 2009; Barjatya & Merritt, 2018; Hoang, Røed, et al., 2018).

In the following, we assess different techniques to infer plasma densities, and satellite potentials from fixed bias needle probe measurements based on synthetic data obtained from kinetic simulations. We also present a new method to interpret m-NLP measurements based on multivariate regression. Our kinetic simulation approach, the con-

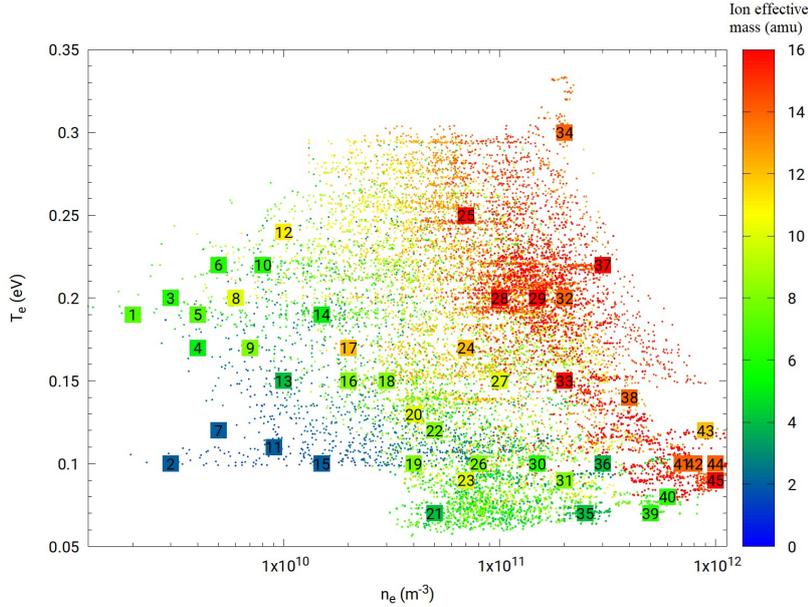


Figure 1. Scatter plot of plasma parameters obtained from the IRI model, corresponding to different latitudes, longitudes, altitudes, and times, as listed in Table 1. The x and y axes, and the color bar refer respectively, to the electron density, electron temperature, and the ion effective mass. Numbered squares identify the set of parameters used in the kinetic simulations.

121 construction of a synthetic data set, and different models to infer plasma parameters are pre-
 122 sented in Sec. 2. In Sec. 3, the various models are assessed using the synthetic data set.
 123 In Sec. 4, the same models are applied to NorSat-1 data, to infer densities and satellite
 124 potentials from in situ measured currents. Section 5 summarizes our findings and presents
 125 some concluding remarks.

126 2 Methodology

127 In this section, we briefly describe our kinetic simulation approach, and how it is
 128 used to construct synthetic data sets used to train and validate inference models, using
 129 two regression techniques. We then describe the various models to infer density, and satel-
 130 lite potential from m-NLP measurement.

131 2.1 Kinetic simulations

132 The space plasma parameters considered in our simulations are selected so as to
 133 be representative of conditions expected for a satellite in low Earth orbit at altitudes rang-
 134 ing between 550 and 650 km. This is done by sampling ionospheric plasma parameters
 135 using the International Reference Ionosphere (IRI) (Bilitza et al., 2014) model in a broad
 136 range of latitudes, longitudes, altitudes, and times as shown in Fig. 1. The ranges con-
 137 sidered for these parameters are summarized in Tab. 1. Forty-five sets of plasma param-
 138 eters approximately evenly distributed in this parameter space are selected as input in
 139 simulations, as shown in numbered squares in Fig. 1. The three-dimensional PIC code
 140 PTetra (Marchand, 2012; Marchand & Lira, 2017) is used to simulate probe currents in
 141 this study. Cross comparisons are made between PTetra simulation results and analytic
 142 results under conditions when those are valid, and with other independently developed
 143 simulation codes, and show excellent agreement (Deca et al., 2013; Marchand et al., 2014).

144 In PTetra, space is discretized using unstructured adaptive tetrahedral meshes (Frey &
 145 George, 2007; Geuzaine & Remacle, 2009). Poisson's equation is solved at each time step
 146 using Saad's GMRES sparse matrix solver (Saad, 2003) in order to calculate the elec-
 147 tric field in the system. Then, electron and ion trajectories are calculated kinetically us-
 148 ing their physical charges and masses self consistently. The mesh for the m-NLP and the
 149 simulation domain illustrated in Fig. 2, is generated with GMSH (Geuzaine & Remacle,
 150 2009). The needle probe used in the simulation has a length of 25 mm and a diameter
 151 of 0.51 mm, as those on the NorSat-1. The needle probe is attached to a 15 mm long
 152 and 2.2 mm diameter guard which is biased to the same voltage as the probe. The outer
 153 boundary of the simulation domain is closer to the probe on the ram side, and farther
 154 on the wake side, as shown in Fig. 2. The probes are assumed to be sufficiently far on
 155 the ram side, away from other satellite components, to be unaffected by their presence,
 156 and are identical except for their biases. A single probe is simulated at a time, and a syn-
 157 thetic solution library is then constructed by simulating the probe under different plasma
 158 conditions and voltages as described in Sec. 2.2. The simulations are made using two
 159 different domain sizes depending on the Debye length of the plasma. For plasma den-
 160 sities below $2 \times 10^{10} \text{ m}^{-3}$ corresponding to a Debye length of 1.9-7.2 cm, a larger do-
 161 main is used. For plasma densities above $2 \times 10^{10} \text{ m}^{-3}$, corresponding to a Debye length
 162 of 0.2-2.2 cm, a smaller domain with finer resolution is used. The simulation size, the
 163 resolution, the number of tetrahedra, and the corresponding Debye length are summa-
 164 rized in Tab. 2. There is overlap between the two simulation domains for simulations
 165 with Debye lengths around 2 cm. No obvious difference was found in the simulated cur-
 166 rents, indicating that simulation results from both domains are consistent in the tran-
 167 sition range. Simulation results from both domains are included when training the re-
 168 gression models. All simulations start initially with 100 million ions and electrons, but
 169 these numbers vary through a simulation, due to particles being collected, leaving, or en-
 170 tering the domain. In the simulations, the probe is segmented into five segments of equal
 171 lengths as shown in Fig. 2, making it possible to estimate a rough distribution of the cur-
 172 rent along its length. The current used to build regression models is a sum of the cur-
 173 rents of the five different segments. The orbital speed of the satellite is assumed to be
 174 fixed at 7500 m/s in the simulations, with a direction perpendicular to the probe. For
 175 the voltages considered, probes are expected to collect mainly electron currents. For sim-
 176 plicity, only two types of ions are considered in the simulation, O^+ and H^+ ions, and
 177 no magnetic field is accounted for, which is justified by the fact that the Larmor radius
 178 of the electron considered is much larger than the radius of the probe. NorSat-1 satel-
 179 lite has a Sun-synchronous orbit, thus moving approximately parallel to the magnetic
 180 field near the equator. As a result, in these regions $\vec{V} \times \vec{B}$ should be small at low and
 181 mid magnetic latitudes, and it is not accounted for in the simulations.

182 2.2 Synthetic solution library

183 In order to assess the inference skill of a regression model, a cost function is de-
 184 fined with the following properties: i) it is non-negative, ii) it vanishes if model inferences
 185 agree exactly with known data in a data set, and iii) it increases as inferences deviate
 186 from actual data. The cost functions used in this work are: the root mean square error,

$$187 \quad RMS = \sqrt{\frac{1}{N_{data}} \sum_{i=1}^{N_{data}} (Y_{mod_i} - Y_{data_i})^2}, \quad (2)$$

188 the root mean square relative error

$$189 \quad RMSr = \sqrt{\frac{1}{N_{data}} \sum_{i=1}^{N_{data}} \frac{(Y_{mod_i} - Y_{data_i})^2}{Y_{mod_i}^2}}, \quad (3)$$

190 the maximum absolute error

$$191 \quad MAE = \max \{|Y_{mod} - Y_{data}|\}, \quad (4)$$

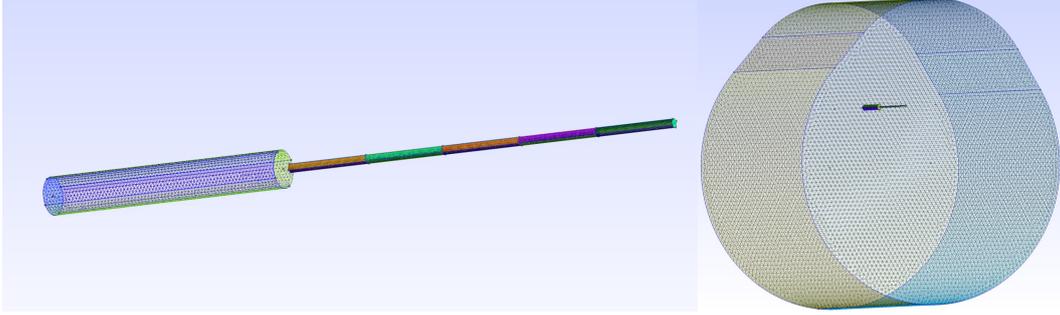


Figure 2. Illustration of an m-NLP geometry (left), and the simulation domain (right). The needle probe has a length of 25 mm and a radius of 0.255 mm, with a guard of 15 mm in length and 1.1 mm in radius. The ram flow is from the top of the simulation domain and is assumed to be 7500 m/s.

Table 1. Spatial and temporal parameters used to sample ionospheric plasma conditions in IRI, and the corresponding ranges in space plasma parameters.

Environment and plasma conditions	Parameter range
Years	1998 2001 2004 2009
Dates	Jan 4 Apr 4 Jul 4 Oct 4
Hours	0-24 with increment of 8 hours
Latitude	-90° - $+90^\circ$ with increment of 5°
Longitude	0° - -360° with increment of 30°
Altitude	550-650 km with increment of 50 km
Ion temperature	0.07-0.16 eV
Electron temperature	0.09-0.25 eV
Effective ion mass	2-16 amu
Density	$2 \times 10^9 - 1 \times 10^{12} \text{m}^{-3}$

Table 2. Parameters used in the two simulation domains are listed. The first two columns give the distances between the probe to the outer boundary on the ram side (D_{ram}), and the wake side (D_{wake}) respectively, followed by the simulation resolutions at the probe, guard, and the outer boundary. The number of tetrahedra used in the simulations is in the order of millions. The corresponding range in Debye lengths is also listed.

D_{ram}	D_{wake}	Probe resolution	Guard resolution	Boundary resolution	Tetrahedra	Debye length
3.5 cm	7 cm	51 μm	220 μm	2 mm	2.5 M	0.2-2.2 cm
30 cm	40 cm	51 μm	220 μm	1 cm	1.7 M	1.9-7.2 cm

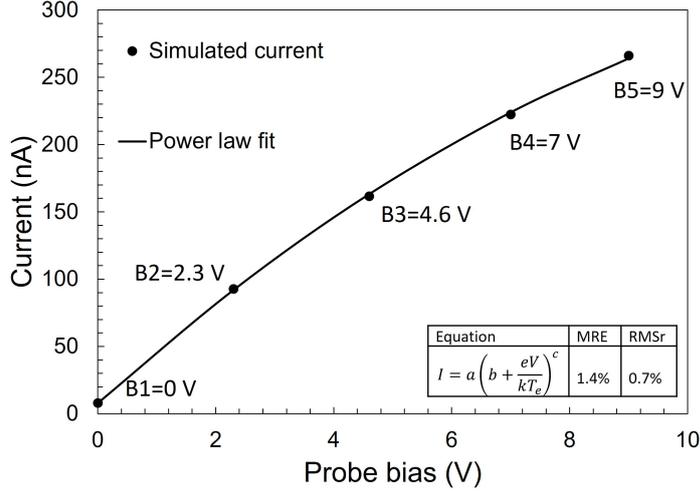


Figure 3. Comparison between calculated currents from PIC simulations, and fitted values using Eq. 6, assuming a density of $2 \times 10^{10} \text{ m}^{-3}$, an effective mass of 8 amu, an electron and ion temperatures of 0.15 and 0.12 eV respectively, corresponding to point 16 in Fig. 1. The fitting errors in the figure are calculated over all 45 sets of plasma conditions using Eq. 3 and 5.

192 and the maximum relative error

$$193 \quad MRE = \max \left\{ \left| \frac{Y_{mod} - Y_{data}}{Y_{mod}} \right| \right\}, \quad (5)$$

194 where Y_{data} and Y_{mod} represent respectively known and inferred plasma parameters, and
 195 N_{data} is the total number of data points.

196 For each of the 45 sets of plasma conditions corresponding to squares in Fig. 1, 5
 197 simulations are made assuming 5 probe voltages (0 V, 2.3 V, 4.6 V, 7 V, 9 V) with re-
 198 spect to background plasma, and the simulated currents vs probe voltage are fitted an-
 199 alytically with:

$$200 \quad I = a \left(b + \frac{eV}{kT_e} \right)^c, \quad (6)$$

201 where a , b , and c are adjustable fitting parameters. The MRE calculated for the 45 fits
 202 is 1.4%, and the RMSr is 0.7%, which shows excellent agreement with simulated collected
 203 currents. A comparison between fitted and computed currents is shown in Fig. 3. The
 204 NorSat-1 m-NLP probes fixed biases V_b are +10, +9, +8, and +6 V, and the probe volt-
 205 age with respect to background plasma is given by the sum of the spacecraft floating po-
 206 tential and the probe bias $V = V_f + V_b$. In simulations, probe currents calculated for
 207 voltages with respect to background plasma in the range between 0 to 9 volts are con-
 208 sidered as shown in Fig. 3. Considering the probe bias voltages V_b given above, probe
 209 currents can be determined, corresponding to arbitrary floating potentials between -1
 210 V and -6 V. A synthetic solution library is created for randomly distributed spacecraft
 211 floating potentials in the range between -1 and -6 V with corresponding currents obtained
 212 by interpolation using Eq. 6 with the fitted a , b , and c computed for each of the 45 cases
 213 considered. The result is a synthetic solution library consisting of four currents collected
 214 by the four needle probes at the four different bias voltages, for 160 randomly distributed
 215 spacecraft potentials in the range between -1 V to -6 V for each of the 45 sets of plasma
 216 parameters. In each entry of the data set, these four currents are followed by the elec-
 217 tron density, the spacecraft potential the electron and ion temperatures, and the ion ef-
 218 fective mass as listed in Tab. 3. The resulting solution library consisting of $45 \times 160 =$
 219 7200 entries is then used to construct a training set with 3600 randomly selected nodes

Table 3. Example entries of the synthetic data set, with currents I_1 , I_2 , I_3 , and I_4 calculated using Eq. 6, and V_b set to 10, 9, 8, and 6 V, respectively. The floating potential V_f is selected randomly in the range of -1 to -6 V, and the probe voltages with respect to background plasma are given by $V = V_b + V_f$. The coefficients, a, b and c are obtained from a nonlinear fit of the simulated currents using Eq. 6. The first and second entries correspond respectively to points 16 and 21 in Fig. 1.

$I_1(nA)$	$I_2(nA)$	$I_3(nA)$	$I_4(nA)$	$V_f(V)$	$n_e(m^{-3})$	$T_e(eV)$	$T_i(eV)$	$m_{\text{eff}}(\text{amu})$
233	208	183	129	-2.50	2×10^{10}	0.15	0.12	8
596	533	467	323	-2.93	5×10^{10}	0.07	0.07	4

220 or entries, and a validation set with the remaining 3600 nodes. The cost functions re-
 221 ported in what follows, used to assess the accuracy of inferences, are all calculated from
 222 the validation data set unless stated otherwise.

223 **2.3 Multivariate regression**

224 In a complex system where the relation between independent variables and depen-
 225 dent variables cannot readily be cast analytically, multivariate regressions based on ma-
 226 chine learning techniques are powerful alternatives to construct approximate inference
 227 models. In this approach, the model must be capable of capturing the complex relation-
 228 ship between dependent and independent variables. Once the model is trained using the
 229 training set, it can then be used to make inferences for cases not included in the train-
 230 ing data set. In this work, two multivariate regression approaches are used to infer plasma
 231 parameters: the Radial Basis Function and Feedforward Neural Networks. The models
 232 are trained by minimizing their cost function on the training data set, and then applied
 233 to the validation data set to calculate the validation cost function without further op-
 234 timization. The use of a validation set is to avoid “overfitting” because there are certain
 235 limitations on the refinement of a model on a training set, such that further improve-
 236 ment of model inference skill in the training set will worsen the model inference skill in
 237 the validation set. A good model is one with the right level of training so as to provide
 238 the best inference skill in the validation set.

239 **2.3.1 Radial basis function**

240 Radial basis function (RBF) multivariate regression is a simple and robust tool used
 241 in many previous studies to infer space plasma parameters using a variety of instruments
 242 with promising results (Liu & Marchand, 2021; Olowookere & Marchand, 2021; Chalaturnyk
 243 & Marchand, 2019; Guthrie et al., 2021). A general expression for RBF regression for
 244 a set of independent n-tuples \bar{X} and corresponding dependent variable Y is given by:

$$245 \quad Y = \sum_{i=1}^N a_i G(|\bar{X} - \bar{X}_i|). \quad (7)$$

246 In general, the dependent variable Y can also be a tuple, but for simplicity, and with-
 247 out loss of generality, we limit our attention to scalar dependent variables. In Eq. 7, the
 248 \bar{X}_i represents the N centers, G is the interpolating function, and the a_i are fitting col-
 249 location coefficients which can be determined by requiring collocation at the centers; that
 250 is, by solving the system of linear equations

$$251 \quad \sum_{i=1}^N a_i G(|\bar{X}_k - \bar{X}_i|) = Y_k \quad (8)$$

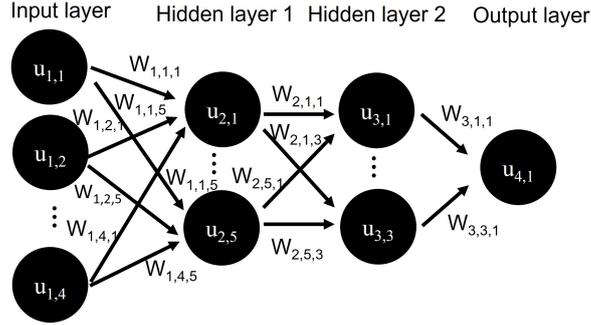


Figure 4. Schematic of a feedforward neural network.

252 for $k = 1, \dots, N$. Here, the dependent variable Y corresponds to the physical param-
 253 eter to be inferred, and the independent variable \bar{X} is a 4-tuple corresponding to the cur-
 254 rents or the normalized currents from the m-NLPs depending on which physical param-
 255 eters are being inferred. There are different ways to distribute the centers in RBF re-
 256 gression. One straightforward approach is to select centers from the training data set,
 257 and evaluate the cost function over the entire training data set for all possible combi-
 258 nations of centers, then select the model which yields the optimal cost function. For this
 259 approach, the number of combinations required for \mathcal{N} data points and N centers is given
 260 by

$$261 \binom{\mathcal{N}}{N} = \frac{\mathcal{N}!}{N!(\mathcal{N} - N)!}. \quad (9)$$

262 This, of course, can be prohibitively large and time-consuming for a large training data
 263 set or using a large number of centers. An alternative strategy is to successively train
 264 models with randomly selected small subsets of the entire training data set using the straight-
 265 forward approach, while calculating the cost function on the full training set, and then
 266 carrying the optimal centers from one iteration to the next. This “center-evolving strat-
 267 egy” is very efficient in finding near-optimal centers for large training data sets and has
 268 proven to be as accurate as the straightforward extensive approach (Liu & Marchand,
 269 2022). The RBF models here follow this procedure. Different G functions and cost func-
 270 tions are tested, and only the models that yield optimal results are reported in this pa-
 271 per.

272 2.3.2 Feedforward neural network

273 The second multivariate regression approach is a Feedforward neural network as
 274 illustrated in Fig. 4. This consists of an input layer, hidden layers, and an output layer.
 275 Each node j in a given layer i in the network is assigned a value $u_{i,j}$, and the node in
 276 the next layer $i+1$ are “fed” from numerical values from the nodes in the previous layer
 277 according to

$$278 u_{i+1,k} = f \left(\sum_{j=1}^{n_i} w_{i,j,k} u_{i,j} + b_{i,k} \right), \quad (10)$$

279 where $w_{i,j,k}$ are weight factors, $b_{i,j}$ are bias terms, and f is a nonlinear activation func-
 280 tion (Goodfellow et al., 2016). In this work, the input layer neurons contain the four-
 281 needle probe currents or normalized currents depending on the physical parameter to
 282 be inferred, whereas the output layer contains one physical parameter. The number of
 283 hidden layers and the number of neurons in the hidden layers are adjusted to fit the spe-
 284 cific problem, and attain good inference skills. The Feedforward neural network is built
 285 using TensorFlow (Abadi et al., 2016) with Adam optimizer (Kingma & Ba, 2015), and
 286 using the ReLU activation function defined as $f(x) = \max(0, x)$. The input variables

are normalized using the `preprocessing.normalization` TensorFlow built-in function which normalizes the data to have a zero mean and unit variance. The structure of the network will be described later when presenting model inferences.

2.4 Space plasma and satellite parameter inference models

The next step is to construct models that map the measured currents to the corresponding plasma and satellite conditions in the solution library. Various models used to infer plasma densities and satellite potentials are described in this section.

2.4.1 Density inference

The density can be inferred directly from the above two multivariate regression models using the currents collected by the four probes as inputs. The density can also be inferred using Eq. 1 which can be rewritten as

$$\frac{n_e}{T_e^{\beta-\frac{1}{2}}} = \sqrt{\frac{\pi^2 m_e}{2A^2 e^3}} \left(\frac{I_1^{\frac{1}{\beta}} - I_2^{\frac{1}{\beta}}}{V_1 - V_2} \right)^\beta. \quad (11)$$

In this equation, subscripts 1 and 2 indicate different probes. A special case of this equation was first proposed by Jacobsen, assuming an infinitely long probe, for which $\beta = 0.5$, resulting in

$$n_e = \sqrt{\frac{\pi^2 m_e}{2A^2 e^3}} \sqrt{\frac{I_1^2 - I_2^2}{V_1 - V_2}}, \quad (12)$$

which gives an expression for the electron density, independently of the temperature (Jacobsen et al., 2010). With currents from more than two probes, the density can be calculated from the slope of the current squared as a function of the bias voltage from a linear least-square fit of all probes (Jacobsen et al., 2010). This will be referred as the ‘‘Jacobsen linear fit’’ (JLF) approach. It is now well known, however, that for finite length probes, with lengths not much larger than the Debye length, β typically ranges between 0.6 and 1. This is the case in particular for the needle probes on NorSat-1 with ratios between probe lengths to Debye length ranging from 0.5 to 12.5. As a consequence, when this method is applied to the solution library, the inferred density is typically three times larger than the density used in the simulation as shown with red boxes in Fig 5. Analytic inferences can be improved by adopting a boosting strategy. With this approach, the less accurate analytic model is used as a first approximation, which is then corrected by applying a more advanced regression technique.

Two boosting strategies are used in this study, consisting of i) an affine transformation, and b) RBF. Considering that the Pearson correlation coefficient R is invariant under an affine transformation, it follows that the offset between two data sets, with a high value of R , can be significantly reduced with a simple affine transformation. To be specific, in this case, the density is first approximated using the JLF approach, and an affine transformation is applied to the natural log of the density as in:

$$\ln(n_e^{\text{affine}}) = a \ln(n_e^{\text{JLF}}) + b. \quad (13)$$

where a and b are determined with a simple least square fit to the known log of the densities in the data set. In the second approach, RBF is used to model the discrepancy between the JLF approximated density and the known densities, and the modeled discrepancy is used to correct the first JLF estimate.

The nonlinear least square fit proposed by Barjatya (BNLF) is also used to infer the density and the satellite potential. In their paper, Barjatya, et al. (Barjatya et al., 2009) apply this method to a full characteristic, covering the ion saturation, the electron retardation, and electron saturation regions. This enabled them to infer all four parameters in Eq. 1, namely, n_e , T_e , V_f , and β . In our analysis, inferences are made from only

332 four currents from four probes at fixed bias voltages, all in the electron saturation re-
 333 gion. As shown by Barjatya and Merritt (Barjatya & Merritt, 2018), however, it is dif-
 334 ficult to infer the temperature using this approach, owing to the weak dependence of col-
 335 lected currents on the electron temperature (see Eq. 11). A solution, proposed in (Barjatya
 336 & Merritt, 2018; Hoang, Røed, et al., 2018), then consists of estimating the electron tem-
 337 perature from other measurements, or from the IRI model, and performing the fit for
 338 the remaining three parameters. This simplification is justified by the fact that, follow-
 339 ing this procedure, a 50% error in the temperature, still produces acceptable results for
 340 the other parameters (Barjatya & Merritt, 2018). When applying the BNLF method in
 341 the comparisons below, We assume a fixed electron temperature (~ 3000 K) in our fit,
 342 and then use the 4-tuples of currents using V_f , n_e , and β values as fitting parameters.
 343 The choice of electron temperature also affects the performance of the model. This choice
 344 of temperature (~ 3000 K) is justified by the fact that it gives the best estimates of den-
 345 sity and potential when BNLF is applied to our synthetic data set. In their original ap-
 346 proach, Barjatya et al. assumed half of the probe on the ram side collects electrons due
 347 to the wake effect (Barjatya et al., 2009). With this assumption, they found that the in-
 348 ferred electron densities were more consistent with their inferred ion densities. In the sim-
 349 ulation approach, the wake and its effect on electron collection are accounted for self-
 350 consistently, and excellent inferences can be made without this assumption.

351 **2.4.2 Analytic estimate of V_f**

352 The satellite potential can be inferred directly from the currents using RBF regres-
 353 sion. In this approach, the four currents are normalized by dividing every current by their
 354 sum, in order to remove the strong density dependence on the currents. A neural net-
 355 work does not produce satisfactory in this case, and it is not used to infer the satellite
 356 potential. The floating potential of the spacecraft can also be inferred using the OML
 357 equation, by rewriting equation 1 as:

$$358 \quad V_f \approx V_f + \frac{kT_e}{e} = \frac{V_2 I_1^{\frac{1}{\beta}} - V_1 I_2^{\frac{1}{\beta}}}{I_2^{\frac{1}{\beta}} - I_1^{\frac{1}{\beta}}} = \frac{V_3 I_2^{\frac{1}{\beta}} - V_2 I_3^{\frac{1}{\beta}}}{I_3^{\frac{1}{\beta}} - I_2^{\frac{1}{\beta}}}. \quad (14)$$

359 In this equation, the subscripts 1,2, and 3 refer to different probes, thus there must be
 360 at least three probes in order to solve for β . The bias voltages of the probes and their
 361 corresponding collected currents are known from measurements, thus β can be solved
 362 using a standard root finder. Given β , equation 14 then provides a value for $V_f + \frac{kT_e}{e}$.
 363 In this expression, $\frac{kT_e}{e}$ is the electron temperature in electron-volt, which in the lower
 364 ionosphere at mid latitudes, is of order 0.3 eV or less. Thus, considering that $\frac{kT_e}{e}$ is gen-
 365 erally much smaller than satellite potentials relative to the background plasma, any of
 366 the two terms on the right side of Eq. 14 provides a first approximation of V_f (Guthrie
 367 et al., 2021). This will be referred to as the ‘‘adapted OML’’ approach.

368 **3 Assessment with synthetic data**

369 In this section, we assess our models using synthetic data, which allows us to check
 370 the accuracy, and quantify uncertainties in our inferences. A consistency check strategy
 371 is also introduced to further assess the applicability of our models.

372 **3.1 Density and satellite potential inference**

373 Direct RBF regression is applied to infer the density using the four currents as in-
 374 put variables. When constructing an RBF model with $G(x) = |x|$, minimizing MRE,
 375 and using 6 centers, the RMSr and MRE calculated on the validation data set are 17%
 376 and 35%, respectively. A test is made to infer the density using RBF with 35 randomly
 377 selected entries from the 45 plasma conditions in the solution library. With 30 voltages,

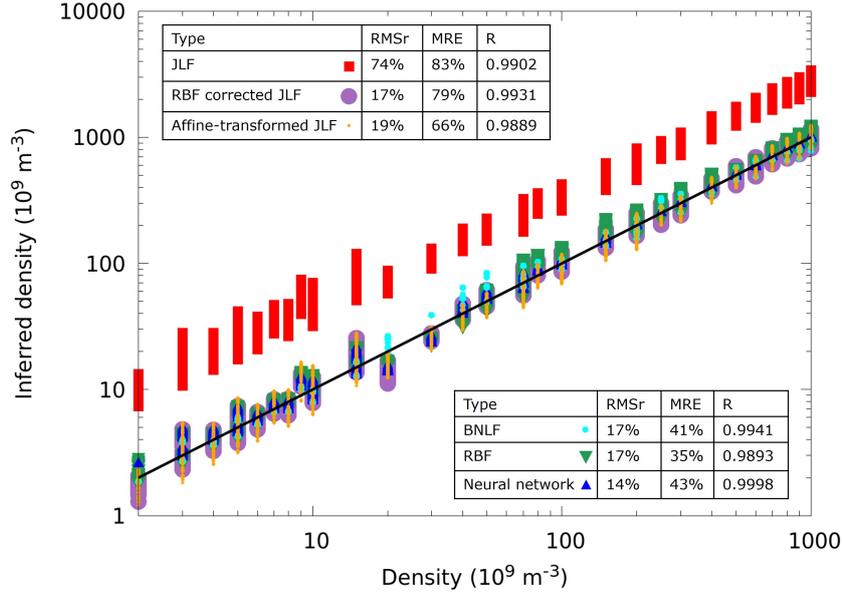


Figure 5. Correlation plot for the density inferences made with different techniques applied to our synthetic validation set. The Pearson correlation coefficient R is calculated using the inferred densities and the density used in the simulation. The black line represents the idealized perfect correlation line.

378 and the same G function, cost function, and number of centers, the calculated MRE is
 379 also 35%. This is an indication that 45 sets of plasma conditions and 160 voltages should
 380 be sufficient in terms of sampling size to construct regression models. Using a neural net-
 381 work with 4 nodes in the input layer, 14 nodes and 12 nodes in two hidden layers, and
 382 1 node in the output layer, results in a 14% RMSr and 43% MRE for the inferred den-
 383 sities. This is calculated using TensorFlow with ADAM optimizer with a learning rate
 384 of 0.005 and an RMSr as the cost function. The input layer is normalized to have a zero
 385 mean and unit variance, while the output layer is normalized by dividing by the largest
 386 density. The densities calculated using the synthetic solution library, as well as the cost
 387 function are shown in Fig. 5.

388 When using an affine transformation to boost the JLF method, the coefficients a
 389 and b in Eq. 13 are obtained from a least-squares fit of the log of these densities, to those
 390 in the training data set. The fitting coefficients in this case, $a = 1.13261$ and $b = -4.82735$,
 391 are then used to perform an affine transformation on the validation data set, leading to
 392 a significant improvement in RMSr from 74% to 19%, and in MRE from 83% to 66% com-
 393 pared to densities inferred from the JLF approach, as shown in Fig 5. When boosting
 394 JLF density with RBF, the 4-tuple of currents is used as input variable \bar{X} . Minimizing
 395 the MRE using $G(x) = |x|$, and 5 centers, the RBF corrected JLF density yields an RMSr
 396 of 17% and an MRE of 79%. The cost functions of the two boosting methods are com-
 397 parable, but an obvious advantage of using an affine transformation is its simplicity.

398 The Python 3 LMFIT package is used to do the nonlinear fit for the BNLf approaches
 399 as in (Debchoudhury et al., 2021). In the fits, the initial values for the density, the po-
 400 tential, and the β value are $8 \times 10^{10} \text{ m}^{-3}$, -3 V and 0.85, and the lower and upper bounds
 401 are 1×10^8 to $1 \times 10^{12} \text{ m}^{-3}$, -6 to -1 V , and 0.49 to 0.99, respectively. The tolerance
 402 of the fit is set to $\text{ftol}=1\text{e-}90$, and the maximum number of function evaluations before
 403 termination is set to $\text{max_nfev}=100000$, to ensure a sufficient number of evaluations be-
 404 fore termination. The potential lower bound of -6 V is needed to ensure that the values
 405 under exponent in Eq. 1 are positive. We obtain 3600 fits for each of the 3600 entries

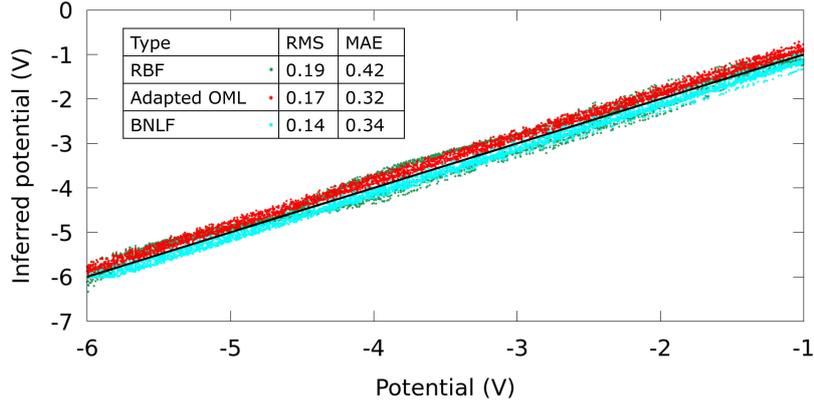


Figure 6. Correlation plot obtained for satellite potential inferred with RBF and OML techniques.

406 of four currents in our validation data set. The overall RMSr calculated using Eq. 3 for
 407 the 3600×4 currents is 1.4 %, and the MRE is 3.8%. The resulting density inferences
 408 have an RMSr of 17 % and an MRE of 41 %, which is better than the densities inferred
 409 from the affine-transformed JLF approach, but less accurate than those from the mul-
 410 tivariate regression models. The β values calculated are in the range of 0.75 to 1. Us-
 411 ing LMFIT and multiprocessing packages, and 10 parallel processors in the Pool, 3600
 412 fits can be done in 0.96 seconds using an AMD 5800x processor. In comparison, linear
 413 fits of the currents square, followed by an affine transformation of the log of the inferred
 414 density can be done using fixed formulas (7400 sets can be fitted in one second using an
 415 AMD 5800x without parallelization), and thus are considerably faster than a nonlinear
 416 fit. Regression methods such as RBF or neural networks are also numerically very ef-
 417 ficient, considering they involve simple arithmetic expressions with pre-calculated coef-
 418 ficients. Compared to the other density models considered, straightforward RBF yields
 419 the smallest MRE, thus it is the preferred model to infer density in this work. However,
 420 the affine-transformed JLF method enables density inferences with accuracy compar-
 421 able to those of more complex approaches. This simple and practical technique should there-
 422 fore be of interest in routine data analysis.

423 When the adapted OML approach is used to infer satellite potentials, an MAE of
 424 0.3 V is calculated using currents collected with probe biases of 10, 9, and 8 volts probes.
 425 Referring to Eq. 14, the error of 0.3 V is likely due in part to the maximum electron tem-
 426 perature of 0.3 eV considered in the simulations. The β values calculated in the synthetic
 427 solution library are in the range of 0.75 to 1. RBF regression is also used to infer satel-
 428 lite potentials. In this case, using $G(x) = |x|$, 5 centers, and minimizing the MAE, the
 429 calculated MAE on the validation data set is 0.42 V, and the RMS is 0.19 V. The in-
 430 ferred satellite potential from the BNLf approach has an RMS of 0.14 V, and an MAE
 431 of 0.34 V, which proves this method to be the most accurate compared to the other meth-
 432 ods considered. A correlation plot for potentials inferred using the RBF, adapted OML,
 433 and BNLf approaches is shown in Fig. 6. All methods show good agreement with val-
 434 ues from the synthetic solution library.

435 3.2 Consistency check

436 In order to further assess the applicability of our inference approaches, we perform
 437 the following consistency check. First, RBF models $M1(n_e)$ and $M1(V_f)$ are constructed
 438 to infer the density and satellite potential using 4-tuple currents from our synthetic data
 439 set. A second model ($M2$) is constructed to infer collected currents from densities and

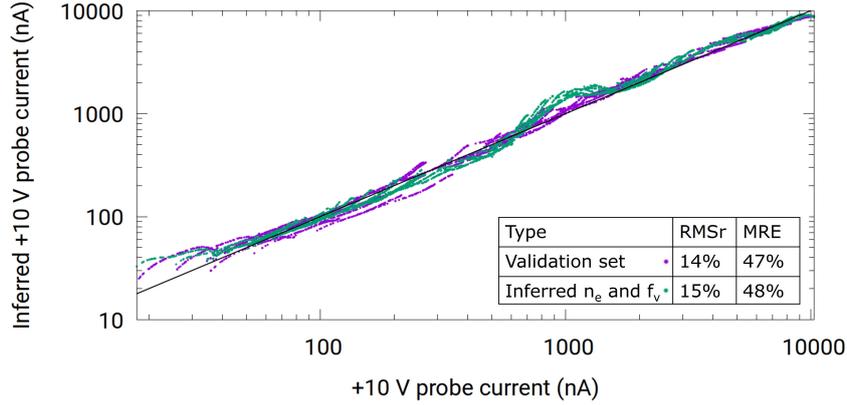


Figure 7. Correlation plot of inferred +10 V probe current against +10 V probe current from the synthetic data set. The calculated +10 probe currents in the purple curve are calculated using the validation data set, while the green curve is calculated using inferred densities and floating potentials from RBF regression.

440 floating potentials in our synthetic data set. Since we are not able to infer temperatures
 441 from the currents, the temperature is not included in $M2$. Consistency is then assessed
 442 in two steps, by i) using currents from synthetic data and models $M1(n_e)$ and $M1(V_f)$
 443 to infer densities and floating potentials, and ii) applying models $M2$ to these inferred
 444 values, to infer back collected currents. RBF density and floating potential inferences
 445 are used in $M1(n_e)$, and $M1(V_f)$ as described in sec. 2.4. RBF is also used in $M2$ with
 446 $G(x) = \sqrt{1 + x^{2.5}}$, and minimizing RMSr with 5 centers. With perfect inference mod-
 447 els, the results for these back-inferred currents, should agree exactly with the starting
 448 currents from synthetic data. Variances between back-inferred and simulated currents
 449 in the synthetic data are presented as indicative of the level of confidence in our regres-
 450 sion techniques. The correlation plot in Fig. 7, shows back-inferred currents (green) cal-
 451 culated for a probe with 10 V bias against known currents from synthetic data. For com-
 452 parison, the figure also shows the correlation between directly inferred currents (purple)
 453 when model $M2$ is applied to densities and floating potentials in the synthetic data set.
 454 Both back-inferred and directly inferred currents are in excellent agreement with known
 455 currents from synthetic data, with comparable metric skills of $\simeq 15\%$ and $\simeq 48\%$ for
 456 the RMSr and the MRE, respectively. Considering that errors are compounded between
 457 the first and second models for the back-inferred currents, the nearly identical metric skills
 458 in Fig. 7 is seen as confirmation of the validity of our regression models.

459 4 Application to NorSat-1 data

460 In this section, we apply our density and potential inference models constructed
 461 with synthetic data, to in situ measurements made with the m-NLP on the NorSat-1 satel-
 462 lite. The NorSat-1 currents were obtained from a University of Oslo data portal (Hoang,
 463 Clausen, et al., 2018). The epoch considered corresponds to one and a half orbit of the
 464 satellite starting at approximately 10:00 UTC on January 4, 2020. We start with a com-
 465 parison of simulated and measured currents to verify that our simulated currents are in
 466 the same range as those of measured in situ currents. Inferences made with RBF, neu-
 467 ral network, BNLF, adapted OML, and the two corrected JLF approaches constructed
 468 in Sec. 2.4, are also presented.

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4.1 Measured in-situ, and simulated currents

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The relevance of the space plasma parameter range considered in the simulations, to NorSat-1, is assessed in Fig. 8, by plotting currents collected by the +9 V probe against that collected by the +10 V, from both synthetic data, and in situ measurements. The close overlap, and the fact that the range of in situ measurements is within the range of simulated currents, indicate that the physical parameters selected in the simulations, are indeed representative of conditions encountered along the NorSat-1 orbit.

The current measurement resolution for the NorSat-1 m-NLP probes is approximately 1 nA (Hoang, Clausen, et al., 2018). The noise level from the environment, however, is estimated to be of order 10 nA. In what follows, darker colors are used to represent inferences made using currents above 10 nA, and lighter colors are used to represent inferences using currents between 1 to 10 nA. This is done by filtering out all data that contain a current that is below 10 nA or 1 nA in any of the four probes. A word of caution is in order, however, for inferences made from these lower currents, as a conservative estimate of the threshold for sufficient signal-to-noise ratios, is approximately 10 nA. This lower bound current is supported by a consistency check made with models 1 and 2 described in Sec. 3.2, and presented below in Sec. 4.3.

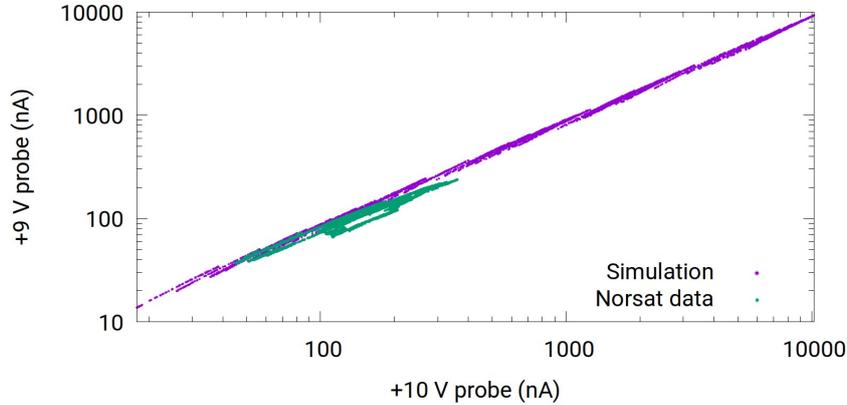


Figure 8. Correlation plot between currents collected by the +9 V and the +10 V probes for both NorSat-1, and synthetic data.

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4.2 Density and satellite potential inference

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Our models, trained with synthetic data as described in Sec. 2.4, are now applied to infer plasma densities and satellite potentials from in situ measured currents, for the time period considered. The results obtained with the different models presented in Sec. 2.4 are shown in Fig. 9 for the inferred densities, satellite potentials, and measured currents collected by the four probes. The position of the satellite relative to the Earth and the Sun given by the solar zenith angle is also plotted in the figure. For example, a small solar zenith angle means that the satellite is near the equator on the dayside.

Applying the BNLF method with only four probes at fixed bias voltages, all in the electron saturation region, is more challenging than applying the technique to a probe operated in sweep mode, covering the ion saturation, the electron retardation, and the electron saturation regions. The reason is that in sweep mode, characteristics contain much more information than in fixed bias mode, with only four probes. In practice, inferences made from sweep mode characteristics are less sensitive to random errors in the currents, which, owing to their larger numbers, tend to cancel. With only four currents,

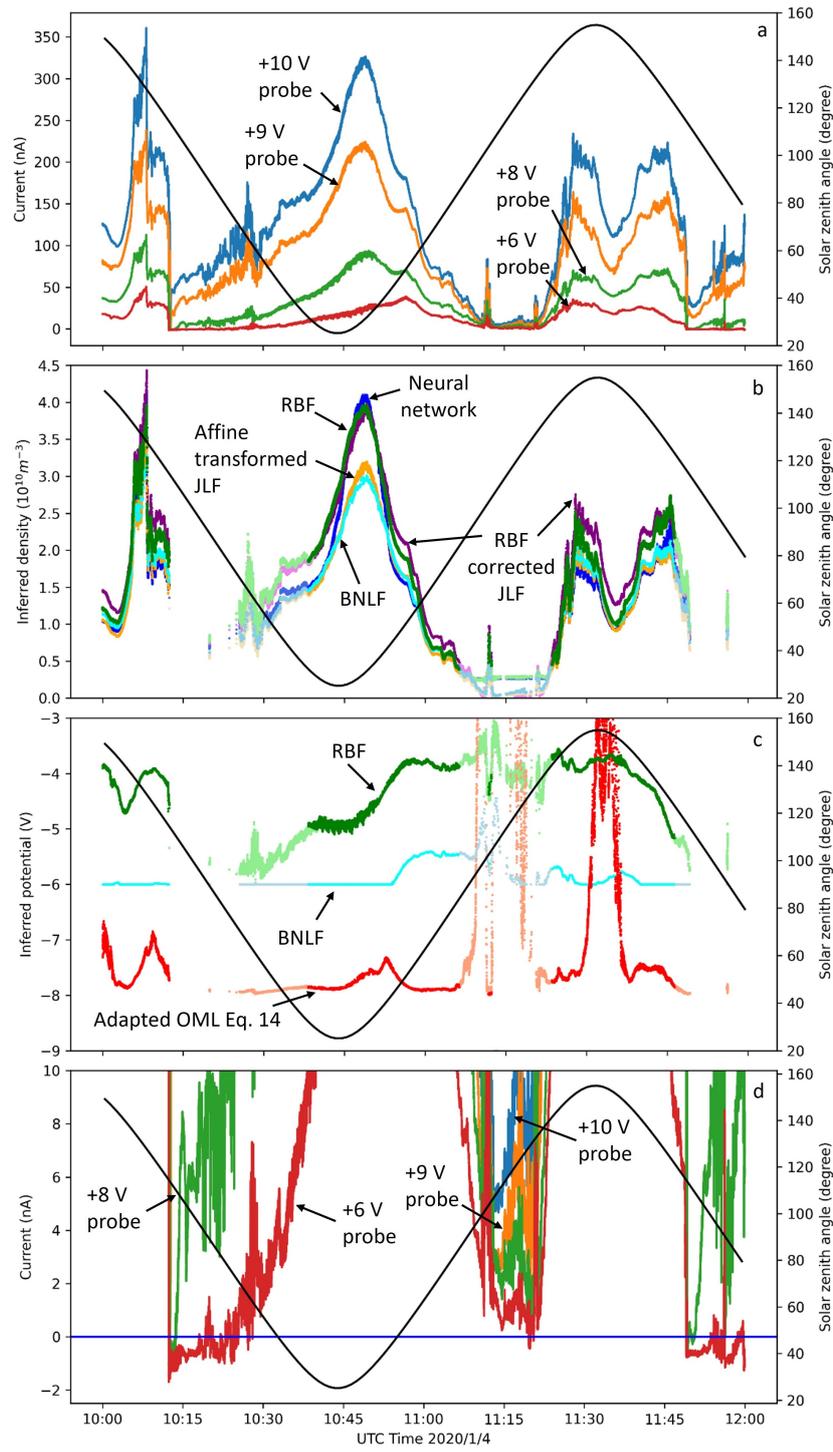


Figure 9. Illustrations of NorSat-1 collected currents considered in this study in panel a, inferred densities in panel b, inferred potentials in panel c, and the NorSat-1 current near 0 A in panel d. The solar zenith angle is also plotted against the secondary axis. Curves in darker colors are from model inferences using data above 10 nA, whereas those in lighter colors show inferences using data with currents between 1 nA and 10 nA.

501 however, noise is less likely to cancel, and inferences will be more sensitive to errors or
 502 noises, in measured currents. For example, the +8 V NorSat-1 probe currents are often
 503 slightly lower than expected for a downward concavity in I as a function of V_b , and tend
 504 to produce an upward concavity with β larger than 1. Thus in most cases, the fitted β
 505 value reaches the maximum allowed value of 1. The resulting inferred densities and satel-
 506 lite potentials are shown in Fig. 9. For the reasons mentioned above, it is clear that no
 507 satellite potential below the fitting lower bound of -6 V can appear in the plot.

508 The densities shown in Fig. 9 panel b are obtained using the five density inference
 509 methods mentioned in Sec. 2.4.1. At 10:45, the neural network density, the RBF cor-
 510 rected JLF density, and the RBF density overlap nicely, while the affine transformed JLF
 511 density and the BNLF density are smaller than other inferred densities, particularly near
 512 the density maxima. The density inferences nonetheless qualitatively agree with each other.
 513 Note that at around 11:15, the inferred densities fall below $2 \times 10^9 \text{ m}^{-3}$, which is out-
 514 side the range of the synthetic solution library. As a result, the regression models such
 515 as RBF and neural networks do not produce the right densities in these ranges. A note
 516 of caution should also be placed for other models at these lighter color regions since the
 517 signal-to-noise ratio is low for data with currents below 10 nA.

518 Using the +10, +9, and +8 NorSat-1 probe currents and Eq. 14, the inferred satel-
 519 lite floating potential is about -8 V for most of the data range considered in this study
 520 as shown in Fig. 9 panel c. This is in stark contradiction with observations in Fig. 9 panel
 521 d, which shows that the +6 V biased probe collects net positive electrons during most
 522 of the period considered. Also, there are periods between 10:15 to 10:30, and after 11:45
 523 when the +6 V probe collects ion current (negative), indicating drops in the satellite po-
 524 tential below -6 V. The poor performance of Eq. 14 to infer the satellite potential here,
 525 results from the fact that Eq. 14 yields erratic values of β ranging from 0.3 to 1.2. At-
 526 tempts have also been made to approximate the satellite potential with Eq. 14 using a
 527 fixed value of 0.58 and 0.78 for β , also resulting in satellite potentials in the -8 V range,
 528 and no improvement was found. This failure to produce acceptable values of the satel-
 529 lite potential clearly shows that the generalized OML expression in Eq. 14 does not pro-
 530 vide a sufficiently accurate approximation for the currents collected by the NorSat-1 probes.

531 The floating potentials inferred from the BNLF model are systematically lower than
 532 those from RBF and are often bounded by the fitting lower limit of -6 V. This is likely
 533 caused by the fact that the +8 V probe current is often lower than expected for a down-
 534 ward concavity in I as a function of V_b . The RBF inferred floating potential shown in
 535 Fig. 9, is within -4 and -6 V, which is consistent with the observation that the +6 V probe
 536 collects electrons during most of the time period considered. This potential is generally
 537 lower than the potential established by the spacecraft on its own, likely due to the large
 538 number of electrons collected by the positively biased solar panels (Ivarsen et al., 2019).
 539 Interestingly, the inferred satellite potential using currents between 1 and 10 nA (light
 540 color) is seen to join smoothly with the darker color inferences, and to decrease below
 541 -6 V around 10:25, which is consistent with the observation that during that time the
 542 +6 V probe no longer collects electron current. The currents collected by the probes are
 543 determined mostly by the density and the satellite potential, and to a lesser extent, by
 544 the electron temperature. In Fig. 9, the density and floating potential are seen to peak
 545 at around 10:45 and 11:00 respectively. The currents from the +8, +9, and +10 V probes
 546 (green, orange, and blue) peak around 10:45, coinciding with the peak in the plasma den-
 547 sity at this time. Then, as time goes forward to 11:00, the currents of the three probes
 548 decrease, also coinciding with a decrease in plasma density. However, the +6 V probe
 549 (red) current is increasing during these times, possibly due to an increase in floating po-
 550 tential. This increase is captured in the RBF and BNLF inferred potential, but not in
 551 the one derived from adapted OML. Another observation is that the inferred floating po-
 552 tential decreases significantly at 10:15, as the satellite crosses the terminator. On NorSat-
 553 1, the negative terminals of the solar cells are grounded to the spacecraft bus while the
 554 positive side is facing the ambient plasma (Ivarsen et al., 2019). A likely explanation for
 555 the potential drop is that the solar cells facing the ambient plasma get charged positively

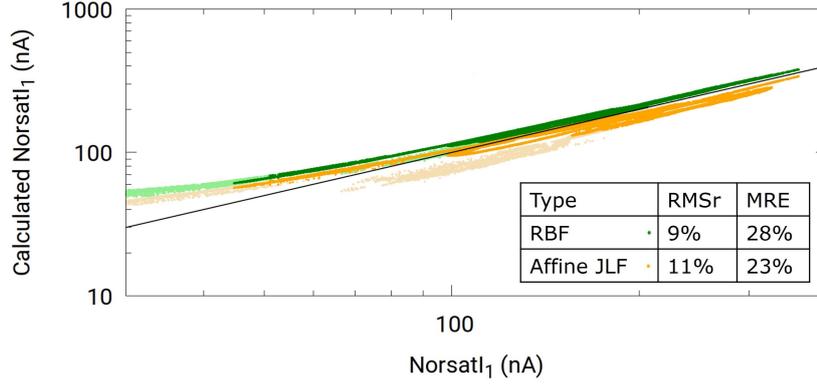


Figure 10. Consistency check is performed in the in situ data following the same procedure as in the synthetic data set. Both models 1 and 2 are trained with our synthetic data, and applied to currents from the +10 V probe on NorSat-1. Darker colors refer to inferences made with currents above 10 nA, while lighter colors refer to inferences obtained with currents between 1 and 10 nA.

556 and suddenly start collecting more electrons upon exiting solar eclipse. This would agree
 557 with findings reported by Ivarsen et al. (Ivarsen et al., 2019).

558 4.3 Consistency check

559 In the absence of accurate and validated inferred densities and satellite potentials
 560 from NorSat-1 data, it is not possible to confidently ascertain to what extent the infer-
 561 ences presented above are accurate. As an alternative, we proceed with a consistency check,
 562 following the same procedure as presented in Sec. 3.2 with synthetic data, but using mea-
 563 sured currents as input. This is done by first applying models $M1(n_e)$ and $M1(V_f)$ trained
 564 with synthetic data, to infer floating potentials and densities from measured currents.
 565 Then $M2$ (also trained with synthetic data) is used to infer currents from the $M1$ - in-
 566 ferred floating potentials and densities. If the models constructed from the synthetic data
 567 also apply to NorSat-1 data, the inferred currents should closely reproduce the measured
 568 NorSat-1 currents. A correlation plot of inferred against measured currents is shown in
 569 Fig. 10 for the +10 V probe. In this plot, the orange and green curves show back-inferred
 570 currents obtained with the RBF $M2$ model. For the orange curve (Affine JLF), the den-
 571 sity used as input in $M2$ is obtained with the affine transformed JLF method. For the
 572 green curve (RBF), the density used as input in $M2$ is obtained with RBF density, while
 573 in both cases, the floating potentials are obtained with the $M1(V_f)$ model using RBF
 574 regression. The parts in lighter colors are obtained using data with a 1nA filter, whereas
 575 the darker color parts are obtained using data with currents above 10 nA. While the graph
 576 only shows currents above 30 nA, the 1 nA filter curve extends to the left down to about
 577 5 nA, however, these calculated +10 volt probe currents plateau in this range and are
 578 far from the measured currents. This behavior is likely due to noise levels of about 10
 579 nA, thus extra caution should be taken when using model inferences for data below 10
 580 nA. The RMSr calculated for the 10 nA NorSat-1 current using direct RBF density as
 581 $M1(n_e)$ is 9%, and the MRE is 28 %, whereas these numbers for the affine transformed
 582 JLF densities are 11 % and 23 %, respectively. The calculated +10 V probe currents based
 583 on RBF regression and affine transformed JLF method nicely follow the measured +10
 584 volt probe current except for a small increase in the variance at lower currents, thus in-
 585 dicating that our model constructed with synthetic data set should be applicable to in
 586 situ data.

587 5 Conclusions

588 Two new approaches are presented and assessed, to infer plasma and satellite pa-
 589 rameters from currents measured with multiple fixed bias needle Langmuir probes. In
 590 the first approach, inferences are made with two multivariate regression techniques, con-
 591 sisting of radial basis functions, and neural networks. The second approach relies on a
 592 simple affine transformation combined with a technique first proposed by Jacobsen to
 593 infer the plasma density. Yet another approach, proposed by Barjatya, et al. is consid-
 594 ered, which consists of performing nonlinear fits of measured currents, to an analytic ex-
 595 pression involving the density, the floating potential, and the exponent β as fitting pa-
 596 rameters, while the electron temperature is estimated by other means. In all cases, the
 597 accuracy of inferences is assessed on the basis of synthetic data obtained from kinetic
 598 simulations made for space-plasma conditions representative of those encountered along
 599 the NorSat-1 satellite. In addition to assessments based on synthetic data, a consistency
 600 check is presented, whereby densities and satellite potentials inferred from collected cur-
 601 rents, are used as input in an inverse regression model to infer currents for one of the
 602 probes. The advantage of this consistency check is that it is applicable to both synthetic,
 603 and in situ measured currents, and in the latter case, it does not rely on a priori given
 604 inferred densities and satellite potentials. Inference consistency checks are made with
 605 both synthetic and in situ measured currents, showing excellent agreement.

606 The density inference methods considered in this study yield excellent results when
 607 applied to the synthetic data set. The models constructed with synthetic data are then
 608 applied to currents measured by the four m-NLP on NorSat-1. The density inferences
 609 from all methods show good agreement, confirming that either method should be a sig-
 610 nificant improvement over the commonly used OML approach based on $\beta = 0.5$. From
 611 our findings, direct RBF and the combination of Jacobsen’s linear fit with $\beta = 0.5$ with
 612 an affine transformation, appear as being the most promising, and deserving of further
 613 study. These two methods provide inferences that are consistent and quantitatively sim-
 614 ilar, while being relatively simple and numerically efficient. The former yields the low-
 615 est maximum relative error when assessed with synthetic data, whereas the latter is the
 616 simplest method and produces inferences with comparable accuracy. The spacecraft float-
 617 ing potential is also inferred using RBF regression, an adapted OML approach, and Bar-
 618 jatya nonlinear fit method. The adapted OML inferences are inconsistent with the mea-
 619 surements from NorSat-1 data since it indicates that the satellite potential is below -6V,
 620 while measurements indicate that the +6 V probe is collecting electron current. Con-
 621 versely, spacecraft potentials inferred with RBF regression are consistent with measured
 622 currents from the +6 V biased probe, showing that the satellite potential must have been
 623 at or above -6 V for most of the one-and-a-half orbital periods considered. This failure
 624 to produce acceptable values of the satellite potential using Eq. 14, and the fact that the
 625 Barjatya nonlinear fit approach with n_e , V_f , and β as fitting parameters, results in β val-
 626 ues appreciably larger than one, shows that in situ measurements on NorSat-1 generally
 627 do not closely follow the empirical expression in Eq. 1. One possible cause of this is that
 628 there might be an offset for the +8 V probe current which is often lower than expected
 629 for a downward concavity in I as a function of V_b . Thus a re-calibration of the instru-
 630 ment, if it were possible, might improve the situation.

631 The analysis presented here has been focused on fixed bias multi-needle Langmuir
 632 probes, with the same dimensions as the ones mounted on NorSat-1, to which it has been
 633 applied as a case study. We stress, however, that the simulation-regression approach to
 634 infer space plasma parameters, is not limited to fixed bias probes or to this particular
 635 configuration of probes. With kinetic solutions capable of reproducing analytic results
 636 under conditions when they are valid, and also capable of accounting for more physics,
 637 and more realistic geometries than theories, solution libraries, training, and validation
 638 sets can just as well be constructed for different probes, mounted on satellites, operated
 639 in fixed or sweep bias voltage mode. By following standard machine learning procedures,
 640 whereby models are trained on a subset of a solution library of known independent and
 641 dependent variables, and tested by applying them to distinct subsets, we can estimate

642 uncertainty margins specifically associated with different inference techniques. Another
 643 important strength of the proposed simulation-regression approach is that it enables rel-
 644 atively straightforward incremental improvements to a model, by accounting for more
 645 physical processes or more detailed geometries; something that would be very difficult
 646 to do in a theory. Implementation of regression models and affine transformation of the
 647 Jacobsen linear fit model involves simple arithmetic expressions with pre-calculated co-
 648 efficients and can easily be programmed for onboard processing of low-level data. These
 649 approaches, however, would require the creation of custom data sets, when applied to
 650 a given mission, so as to account for the geometry relevant to the measuring instruments,
 651 and the space environment conditions expected along a satellite orbit. In cases where
 652 probe characteristics are well approximated with analytic expressions such as Eq. 1, the
 653 BNLF technique should prove fast and convenient, as it does not require extensive sim-
 654 ulations. Custom simulation-regression models, on the other hand, would require more
 655 computational resources, which would necessitate optimization in order to be implemented
 656 onboard a satellite. Despite their complexity, however, such models would have the ad-
 657 vantage of being more general than models based on fits made with empirical analytic
 658 expressions. The work presented here is by no means final. The development of improved
 659 inference approaches based on simulations and regression techniques will require signif-
 660 icantly more efforts, involving collaborations between experimentalists and modelers; an
 661 effort well worth doing, considering the cost and years of preparation involved in scien-
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673 **Data Availability Statement**

674 Simulation data can be accessed through <https://doi.org/10.5281/zenodo.7508167>

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